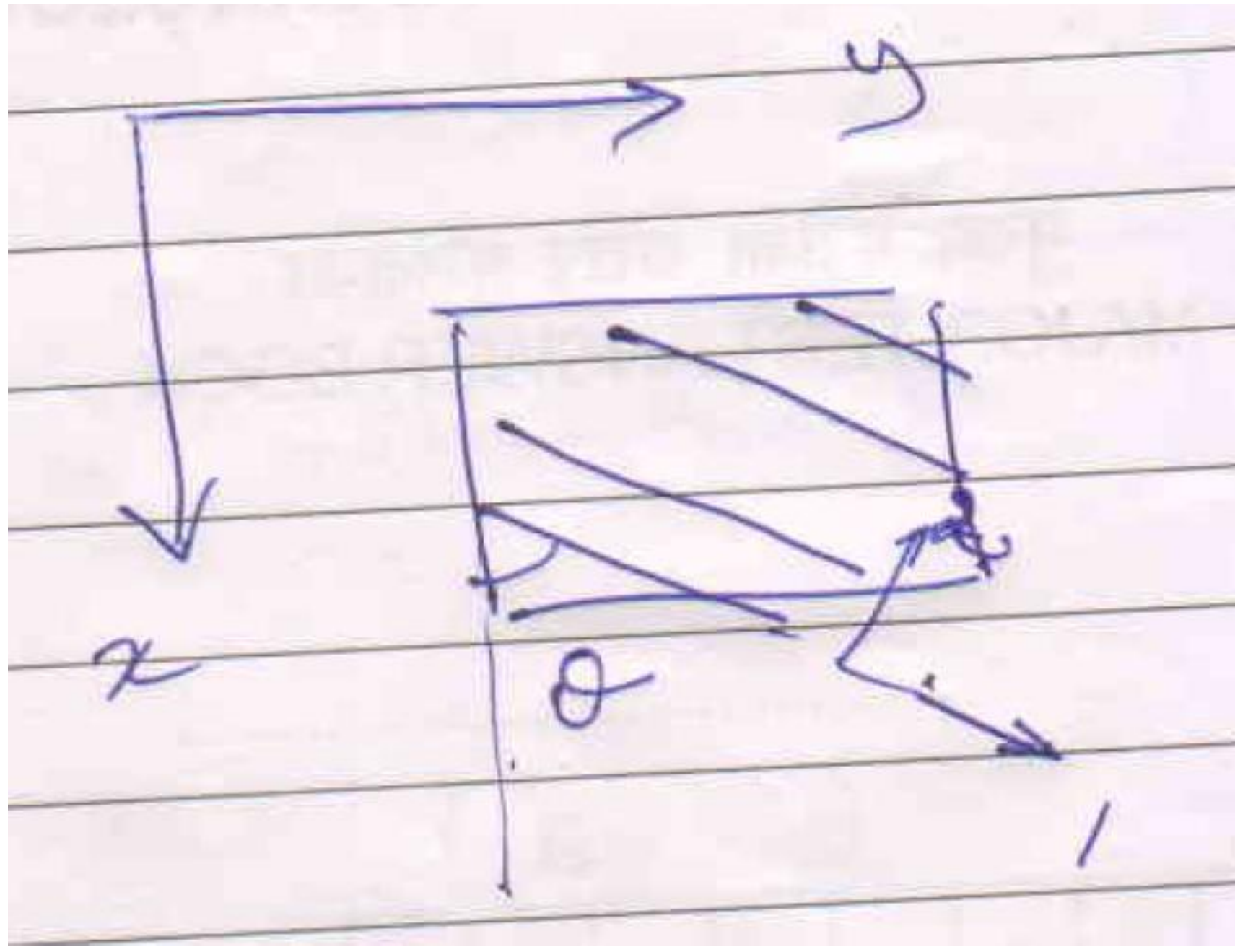


Composites and Thermal Loads



For transversely isotropic

$$G_{23} = \frac{2(1+\nu_{23})}{E_2}$$

$$Q_{11} = \frac{E_1}{1-\nu_{12}\nu_{21}}$$

$$Q_{22} = \frac{E_2}{1-\nu_{12}\nu_{21}}$$

$$Q_{12} = Q_{21} = \frac{\nu_{12} E_2}{1-\nu_{12}\nu_{21}}$$

$$Q_{66} = G_{12}$$

$$\begin{Bmatrix} \sigma_1 \\ \sigma_2 \\ \tau_{xy} \end{Bmatrix} = T_1 \begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \end{Bmatrix}$$

$$T_1 = \begin{bmatrix} \cos^2 \theta & \sin^2 \theta & 2 \cos \theta \sin \theta \\ \sin^2 \theta & \cos^2 \theta & -2 \cos \theta \sin \theta \\ -\cos \theta \sin \theta & \sin \theta \cos \theta & \cos^2 \theta - \sin^2 \theta \end{bmatrix}$$

$$\begin{Bmatrix} \epsilon_1 \\ \epsilon_2 \\ \gamma_{12} \end{Bmatrix} = T_2 \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{bmatrix}$$

$$[T_2] = \begin{bmatrix} \cos^2 \theta & \sin^2 \theta & \cos \theta \sin \theta \\ \sin^2 \theta & \cos^2 \theta & -\cos \theta \sin \theta \\ -2\cos \theta \sin \theta & 2\cos \theta \sin \theta & \cos^2 \theta - \sin^2 \theta \end{bmatrix}$$

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \tau_{xy} \end{Bmatrix} = [T_1]^{-1} \begin{bmatrix} Q_{11} & Q_{12} & 0 \\ Q_{12} & Q_{22} & 0 \\ 0 & 0 & Q_{66} \end{bmatrix} [T_2] \begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \tau_{xy} \end{Bmatrix}$$

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \tau_{xy} \end{Bmatrix} = \begin{bmatrix} Q_{xx} & Q_{xy} & Q_{xs} \\ Q_{xy} & Q_{yy} & Q_{ys} \\ Q_{xs} & Q_{ys} & Q_{ss} \end{bmatrix} \begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \tau_{xy} \end{Bmatrix}$$

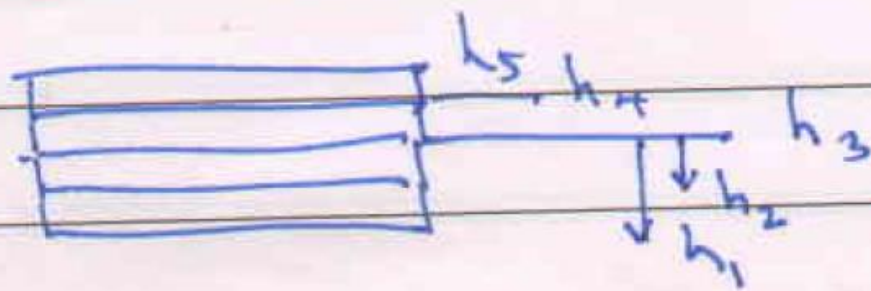
$$\begin{bmatrix} Q_{xx} & Q_{xy} & Q_{xs} \\ Q_{xy} & Q_{yy} & Q_{ys} \\ Q_{xs} & Q_{ys} & Q_{ss} \end{bmatrix} = [T_1]^{-1} \begin{bmatrix} Q_{11} & Q_{12} & 0 \\ Q_{12} & Q_{22} & 0 \\ 0 & 0 & Q_{66} \end{bmatrix} [T_2]$$

$$\begin{Bmatrix} M_x \\ M_y \\ M_{xy} \end{Bmatrix} = \int_{-h/2}^{h/2} \begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{xy} \end{Bmatrix} z \, dz$$

$$= \int_{-h/2}^{h/2} [D] z^2 \begin{Bmatrix} \frac{\partial^2 \phi_x}{\partial x^2} \\ \frac{\partial^2 \phi_y}{\partial y^2} \\ \frac{\partial^2 \phi_x}{\partial y^2} + \frac{\partial^2 \phi_y}{\partial x^2} \end{Bmatrix} dz$$

$$= [D] \begin{Bmatrix} \phi_{x,x} \\ \phi_{y,y} \\ \phi_{x,y} + \phi_{y,x} \end{Bmatrix}$$

$$D = \sum_{k=1}^n [Q]_k \frac{h_k^3 - h_{k-1}^3}{3}$$



$$h_1 = -h/2$$

$$h_5 = h/2$$

$$\begin{Bmatrix} \sigma_{yz} \\ \sigma_{xz} \end{Bmatrix} = \begin{bmatrix} \cos \theta & +\sin \theta \\ -\sin \theta & \cos \theta \end{bmatrix} \begin{Bmatrix} \sigma_{23} \\ \sigma_{13} \end{Bmatrix}$$

$$\begin{Bmatrix} \tau_{yz} \\ \tau_{xz} \end{Bmatrix} = \begin{bmatrix} \cos \theta & +\sin \theta \\ -\sin \theta & \cos \theta \end{bmatrix} \begin{Bmatrix} \tau_{23} \\ \tau_{13} \end{Bmatrix}$$

$$\begin{Bmatrix} \sigma_{yz} \\ \sigma_{xz} \end{Bmatrix} = \begin{bmatrix} \cos \theta & -\sin \theta \\ +\sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} \cos \theta & 0 \\ 0 & \cos \theta \end{bmatrix} \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix} \begin{Bmatrix} \tau_{yz} \\ \tau_{xz} \end{Bmatrix}$$

I rate enthusiasm even above professional skill.

$$\begin{Bmatrix} Q_x \\ Q_y \end{Bmatrix} = \int_{-h/2}^{h/2} \begin{Bmatrix} \sigma_{xz} \\ \sigma_{yz} \end{Bmatrix} dz$$

$$= \sum_{k=1}^N \begin{bmatrix} \cos \tau_{1k} & \sin \tau_{1k} \\ \sin \tau_{1k} & \cos \tau_{1k} \end{bmatrix} \begin{bmatrix} G_{13} & 0 \\ 0 & G_{23} \end{bmatrix} \begin{bmatrix} \cos \tau_{2k} & -\sin \tau_{2k} \\ \sin \tau_{2k} & \cos \tau_{2k} \end{bmatrix}$$

$$\begin{Bmatrix} \sigma_{yz} \\ \sigma_{xz} \end{Bmatrix}$$

- So in the thickness direction Q for each layer is different. The B matrix is same for all layers.
- Once displacement has been calculated stress in a particular element in a layer is calculated using the Q matrix for that layer.

- Thermal Loads for Isotropic Materials

Thermal loads

$$\begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{xy} \end{Bmatrix} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \epsilon_{xx} - \alpha \Delta T \\ \epsilon_{yy} - \alpha \Delta T \\ \gamma_{xy} \end{Bmatrix}$$

$$M = \int \begin{Bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{xy} \end{Bmatrix} z \, dz$$

$$= \int \begin{bmatrix} \phantom{\sigma_{xx}} \\ \phantom{\sigma_{yy}} \\ \phantom{\sigma_{xy}} \end{bmatrix} z \begin{Bmatrix} -z \frac{d^2 \phi}{dx^2} - \alpha \Delta T \\ -z \frac{d^2 \phi}{dy^2} - \alpha \Delta T \\ \gamma_{xy} \end{Bmatrix}$$

$$= \int \left[\begin{array}{c} \text{ } \end{array} \right] z \left\{ \begin{array}{c} -z \frac{d\beta}{dx} \\ -z \frac{d\beta}{dy} \end{array} \right\} \downarrow z$$

$0 \times y$

$-z \left(\frac{\partial \beta}{\partial x} + \frac{\partial \beta}{\partial y} \right)$

$$= \int \left[\begin{array}{c} \text{ } \end{array} \right] z \left\{ \begin{array}{c} -\alpha \Delta T \\ -\alpha \Delta T \\ 0 \end{array} \right\} \downarrow z$$

3×3

3×1

This term goes to RHS and contributes to the thermal load

$$q.w = \int \left[\frac{\partial \delta p_x}{\partial x} \quad \frac{\partial \delta p_y}{\partial y} \quad \frac{\partial \delta p_y}{\partial x} + \frac{\partial \delta p_x}{\partial y} \right]$$

$$\left[\right] z \left\{ \begin{array}{c} -\alpha \Delta T \\ -\alpha \Delta T \\ 0 \end{array} \right\} dz$$

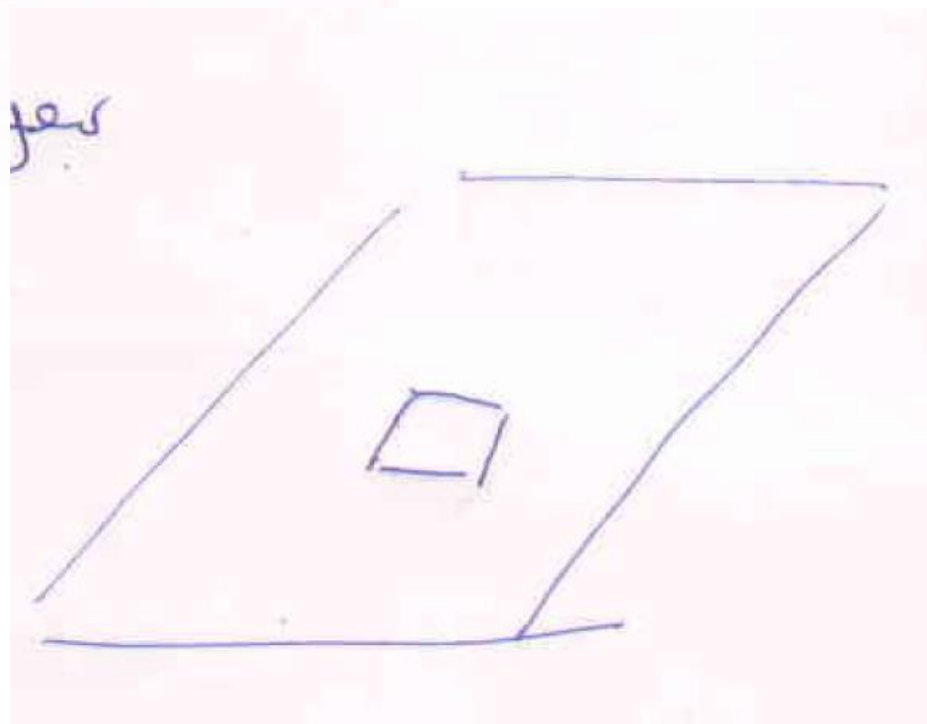
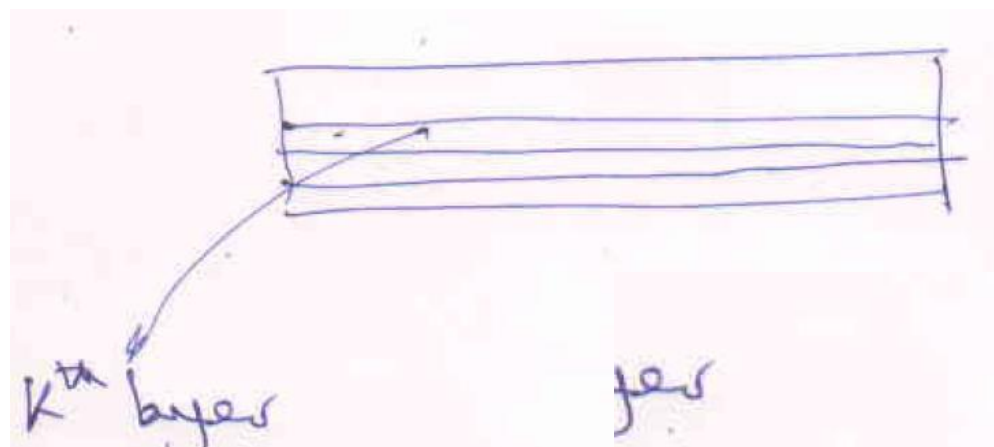
Thermal load

$$\int \mathbf{B}^T \mathbf{B} z \left[\right] \left\{ \begin{array}{c} \alpha \Delta T \\ \alpha \Delta T \\ 0 \end{array} \right\} dz$$

- For composites:

$$\begin{aligned}\alpha_x &= m^2 \alpha_1 + n^2 \alpha_2 \\ \alpha_y &= n^2 \alpha_1 + m^2 \alpha_2 \\ \gamma_{xy} &= 2mn (\alpha_1 - \alpha_2) \\ \alpha_{xz} &= 0 \\ \alpha_{yz} &= 0\end{aligned}$$

$$m = \cos \theta, n = \sin \theta$$



element in
 K^{th} layer

$$\{\sigma_k\}^e = [Q]_k [B]^e \{q\}^e$$

$$\{q\}^e = \begin{Bmatrix} \omega_1 \\ \sigma_{x1} \\ \sigma_{y1} \\ \vdots \\ \omega_4 \\ \sigma_{x4} \\ \sigma_{y4} \end{Bmatrix}^e$$